## Mechanical Engineering

## PAPER-I

## Question Paper Specific Instructions

## Please reach each of the following instruction carefully before attempting questions:

There are EIGHT questions divided in TWO sections.
Candidate has to attempt FIVE questions in all
Questions No. 1 and 5 are compulsory and out of the remaining, any THREE are to be attempted choosing at least ONE question from each section.

The number of marks carried by a question/ part is indicated against it.
Wherever any assumptions are made for answering a question, they must be clearly indicated.

Diagrams/figures, wherever required, shall be drawn in the space provided for answering the question itself.

Unless otherwise mentioned, symbols and rotations carry their usual standard meanings.

Psychometric Chart is given in Page No.8.
Attempts of questions shall be counted in sequential order. Unless struck off, attempt of a question shall be counted even if attempted partly.

Any page or portion of the page left blank in the QCA Booklet must be clearly struck off.
Answers must be written in ENLISH only.

## SECTION-A

1. (a) (i) Differentiate between rotational and irrotationalflows. Can there be any possibility of having zones possessing characteristics of both rotationaland irrotational flows?
(ii) If the expression for the stream function is described by $\psi=x^{3}-3 x y^{2}$ determine whether the flow is rotational or irrotational. Further, find out the correct expression of the velocity potential function of the following two, considering the flow is irrotational:
2. $\phi=y^{3}-3 x^{2} y$
3. $\phi=-7 x^{3} y$
[6 +6 Marks]
(b) A refrigerated truck whose dimensions are $12 \mathrm{~m} \times 2.5 \mathrm{~m} \times 3 \mathrm{~m}$ is to be precooled from $30^{\circ} \mathrm{C}$ to an average temperature of $5^{\circ} \mathrm{C}$. The construction of the truck is such that a transmission heat gain occurs at the rate of $90 \mathrm{~W} /{ }^{\circ} \mathrm{C}$.If the ambient temperature is $30^{\circ} \mathrm{C}$, determine how long it will take for asystem with a refrigeration capacity of 10 kW to precool this truck. The density of air may be taken as $1.2 \mathrm{~kg} / \mathrm{m}^{3}$ and its specific heat at average temperatureof $17.5^{\circ} \mathrm{C}$ is $\mathrm{C}_{\mathrm{p}}=1.0 \mathrm{~kJ} / \mathrm{kg} .{ }^{\circ} \mathrm{C}$. State the assumptions, if any.
[12 Marks]
(c) An engine oil flows through a copper tube of 1 cm internal diameter and 0.02 cm wall thickness at the flow rate of $0.1 \mathrm{~kg} / \mathrm{s}$. Consider that thetemperature of the oil at the entry is $30^{\circ} \mathrm{C}$. If the oil is heated to $50^{\circ} \mathrm{C}$ bysteam condensing at atmospheric pressure, calculate the length of the coppertube. The properties of the oil are as follows:
$\mathrm{C}_{\mathrm{p}}=1964 \mathrm{~J} / \mathrm{kg}-\mathrm{K}, \rho=876 \mathrm{~kg} / \mathrm{m}^{3}, \mathrm{k}=0.144 \mathrm{~W} / \mathrm{m}-\mathrm{K}$,
$\mu=0.210 \mathrm{~N} . \mathrm{s} / \mathrm{m}^{2}, \operatorname{Pr}=2870$
(d) Explain the mechanism of NOx formation and also the methods for its reduction in stationary gas turbine engines.
(e) (i) Why are higher heat transfer rates experienced in drop wise condensation than in film condensation?
(ii) Distinguish between nucleate boiling and film boiling.
4. (a) (i) Find the distance from the pipe wall at which the local velocity is equal tothe average velocity for turbulent flow in pipe.
(ii) Distinguish between hydrodynamically smooth and rough boundaries.
(b) (i) In a closed system, 3 kg of air at initial conditions of 400 kPa and $90^{\circ} \mathrm{Cadiabatically} \mathrm{expands}$ until its volume is 2.5 times the initial volume and temperature becomes equal to that of surroundings. If the conditions of the surroundings are 100 kPa and $25^{\circ} \mathrm{C}$, determine the following for thisprocess:
5. The maximum work
6. The change in availability
7. The irreversibility
(ii) Prove that for an ideal gas, the slope of an isochoric line on theT-s diagram is more than that of the isobaric line.
(c) A square plate heater $(15 \mathrm{~cm} \times 15 \mathrm{~cm})$ is inserted between two slabs. Slab A is 2 cm thick ( $\mathrm{k}=50 \mathrm{~W}$ $/ \mathrm{m}-{ }^{\circ} \mathrm{C}$ ) and slab B is 1 cm thick $\left(\mathrm{k}=0.2 \mathrm{~W} / \mathrm{m}-{ }^{\circ} \mathrm{C}\right)$. The outside heat transfer coefficients on side of A and side of Bare $200 \mathrm{~W} / \mathrm{m}^{2}-{ }^{\circ} \mathrm{C}$ and $50 \mathrm{~W} / \mathrm{m}^{2}-{ }^{\circ} \mathrm{C}$ respectively. The temperature of surrounding air is $25^{\circ} \mathrm{C}$. If the rating of heater is 1 kW , find the
(i) maximum temperature of the system;
(ii) outer surface temperature of two slabs.

Assume steady-state heat flow.
3. (a) A centrifugal pump discharges 2000 litres/s of water developing a head of 20 m when running at $300 \mathrm{r} . \mathrm{p} . \mathrm{m}$. The impeller diameter at the outlet and outlet flow velocity are 1.5 m and $3.0 \mathrm{~m} / \mathrm{s}$ respectively. If the blades are set back at an angle of $30^{\circ}$ at the outlet, determine the-
(i) manometric efficiency;
(ii) power required by the pump;
(iii) minimum speed to start the pump if the inner diameter is 750 mm .
[20 Marks]
(b) Air flows at $12 \mathrm{~m} / \mathrm{s}$ past a smooth rectangular flat plate 0.4 m wide and 3 mlong . Assuming that the transition occurs at $\operatorname{Re}=5.5 \times 10^{5}$, calculate the totaldrag force when
(i) the flow is parallel to the length of the plate;
(ii) the flow is parallel to the width of the plate.

Assume,
Density of air, $\rho=1.24 \mathrm{~kg} / \mathrm{m}^{3}$
Kinematic viscosity, $v=0.15$ stokes
(c) Two tanks, tank A and tank B, are separated by a partition as shown in the figure. Tank A contains 3 kg of steam at 1 MPa and $300^{\circ} \mathrm{C}$. Tank B contains 4 kg of saturated liquid-vapour mixture at $150^{\circ} \mathrm{C}$ with a dryness fraction of0.5. The partition is removed and two fluids are allowed to mix until thethermal equilibrium and mechanical equilibrium are acquired. If the pressure of the final state is 300 kPa ; determine-
(i) the temperature of the final state;
(ii) the quality of the steam at final state;
(iii) the amount of heat lost from the tanks.

[Required steam tables are attached below]
$P=200 \mathrm{kPa}(120.23)$

|  | $P=200 \mathrm{kPa}(120.23)$ |  |  |  | $P=300 \mathrm{kPa}(133.55)$ |  |  |  | $P=400 \mathrm{kPa}$ (143.63) |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $T$ | $v$ | $u$ | h | $s$ | $v$ | $u$ | $h$ | 5 | $v$ | u | $h$ | $s$ |
| 900 | 2.70643 | 3854.5 | 4395.8 | 9.4565 | 1.80406 | 3854.2 | 4395.4 | 9.2691 | 1.35288 | 3853.9 | 4395.1 |  |
| 1000 | 2.93740 | 4052.5 | 4640.0 | 9.6563 | 1.95812 | 4052.3 | 4639.7 | 9.4689 | 1.46847 | 4052.0 | 4639.4 | 9.3360 |
| 1100 | 3.16834 | 4257.6 | 4890.7 | 9.8458 | 2.11214 | 4256.8 | 4890.4 | 9.6585 | 1.58404 | 4256.5 | 4890.1 | 9.5255 |
| 1200 | 3.39927 | 4467.5 | 5147.3 | 10.0262 | 2.26614 | 4467.2 | 5147.1 | 9.8389 | 1.69958 | 4467.0 | 5146.8 | 9.7059 |
| 1300 | 3.63018 | 4683.2 | 5409.3 | 10.1982 | 2.42013 | 4683.0 | 5409.0 | 10.0109 | 1.81511 | 4682.8 | 5408.8 | 9.8780 |
|  | $P=500 \mathrm{kPa}(151.86)$ |  |  |  | $P=600 \mathrm{kPa}(158.85)$ |  |  |  | $P=800 \mathrm{kPa}(170.43)$ |  |  |  |
| Sat. | 0.37489 | 25612 | 2748.7 | 6.8212 | 0.31567 | 2567.4 | 2756.8 | 6.7600 | 0.24043 | 2576.8 | 2769.1 | 6.6627 |
| 200 | 0.42492 | 2642.9 | 2855.4 | 7.0592 | 0.35202 | 2638.9 | 2850.1 | 6.9665 | 0.26080 | 2630.6 | 2839.2 | 6.8158 |
| 250 | 0.47436 | 2723.5 | 2960.7 | 7.2708 | 0.39383 | 2720.9 | 2957.2 | 7.1816 | 0.29314 | 2715.5 | 2950.0 | 7.0384 |
| 300 | 0.52256 | 2802.9 | 3064.2 | 7.4598 | 0.43437 | 2801.0 | 3061.6 | 7.3723 | 0.32411 | 2797.1 | 3056.4 | 7.2372 |
| 350 | 0.57012 | 2882.6 | 3167.6 | 7.6328 | 0.47424 | 2881.1 | 3165.7 | 7.5463 | 0.35439 | 2878.2 | 3161.7 | 7.4088 |
| 400 | 0.61728 | 2963.2 | 3271.8 | 7.7937 | 0.51372 | 2962.0 | 3270.2 | 7.7078 | 0.38426 | 2959.7 | 3267.1 | 7.5715 |
| 500 | 0.71093 | 3128.4 | 3483.8 | 8.0872 | 0.59199 | 3127.6 | 3482.7 | 8.0020 | 0.44331 | 3125.9 | 3480.6 | 7.8672 |
| 600 | 0.80406 | 3299.6 | 3701.7 | 8.3521 | $00_{6} 66974$ | 3299.1 | 3700.9 | 8.2673 | 0.50184 | 3297.9 | 3699.4 | 8.1332 |
| 700 | 0.89691 | 3477.5 | 3926.0 | 8.5952 | 0,74720 | 3477.1 | 3925.4 | 8.5107 | 0.56007 | 3476.2 | 3924.3 | 8.3770 |
| 800 | 0.98959 | 3662,2 | 4157.0 | 8.8211 | 0.82450 | 3661.8 | 4156.5 | 8.7367 | 0.61813 | 3661.1 | 4155.7 | 8.6033 |
| . 900 | 1.08217 | 3853.6 | 4394.7 | 9.0329 | 0.90169 | 3853.3 | 4394.4 | 8.9485 | 0.67610 | 3852.8 | 4393.6 | 8.8153 |
| 1000 | 1.17469 | 4051.8 | 4639.1 | 9,2328 | 0.97883 | 4051.5 | 4638.8 | 9.1484 | 0.73401 | 4051.0 | 4638.2 | 9.0153 |
| 1100 | 1.26718 | 4256.3 | 4889.9 | 9.4224 | 1,05594 | 4256.1 | 4889.6 | 9.3381 | 0.79188 | 4255.6 | 4889.1 | 9.2049 |
| 1200 | 1.35964 | 4466.8 | 5146.6 | 9.6028 | 1.13302 | 4466.5 | 5146.3 | 9.5185 | 0.84974 | 4466.1 | 5145.8 | 9.3854 |
| 1300 | 1.45210 | 4682.5 | 5408.6 | 9.7749 | 1.21009 | 4682.3 | 5408.3 | 9.6906 | 0.90758 | 4681.8 | 5407.9 | 9.5575 |
|  |  | $=1.00$ | (179.91) |  |  | $P=1.2$ | (187. |  |  | $=1.40 \mathrm{M}$ | $\mathrm{Pa}(195.07)$ |  |
| Sat. | 0.19444 | 2583.6 | 2778.1 | 6.5864 | 0.16333 | 2588.8 | 2784.8 | 6.5233 | 0.14084 | 2592.8 | 2790.0 | 6.4692 |
| 200 | 0.20596 | 2621.9 | 2827.9 | 6.6939 | 0.16930 | 2612.7 | 2815.9 | 6.5898 | 0.14302 | 2603.1 | 2803.3 | 6.4975 |
| 250 | 0.23268 | 2709.9 | 2942.6 | 6.9246 | 0.19235 | 2704.2 | 2935.0 | 6.8293 | 0.16350 | 2698.3 | 2927.2 | 6.7467 |
| 300 | 0.25794 | 2793.2 | 3051.2 | 7.1228 | 0.21382 | 2789.2 | 3045.8 | 7.0316 | 0.18228 | 2785.2 | 3040.4 | 6.9533 |
| 350 | 0.28247 | 2875.2 | 3157.7 | 7.3010 | 0.23452 | 2872.2 | 3153.6 | 7.2120 | 0.20026 | 2869.1 | 3149.5 | 7.1359 |
| 400 | 0.30659 | 2957.3 | 3263.9 | 7.4650 | 0.25480 | 2954.9 | 3260.7 | 7.3773 | 0.21780 | 2952.5 | 3257.4 | 7.3025 |
| 500 | 0.35411 | 3124.3 | 3478.4 | 7.7621 | 0.29463 | 3122.7 | 3476.3 | 7.6758 | 0.25215 | 3121.1 | 3474.1 | 7.6026 |


| Steam Table |  |  |  |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\begin{gathered} \text { Temp. } \\ { }^{\circ} \mathrm{C} . \\ T \end{gathered}$ | Pressure <br> $\mathrm{kPa}, \mathrm{MPa}$ $P$ | $\begin{gathered} \text { Sat. } \\ \text { Liquid } \\ v_{\mathrm{f}} \end{gathered}$ | Sat. Vapour $\mathbf{v}_{\mathrm{g}}$ | $\begin{gathered} \hline \text { Sat. } \\ \text { Liquid } \\ u_{\mathrm{f}} \end{gathered}$ | Evap. $u_{f g}$ | $\begin{gathered} \text { Sat. } \\ \text { Vapour } \\ u_{g} \end{gathered}$ | $\begin{gathered} \text { Sat. } \\ \text { Liquid } \\ h_{f} \end{gathered}$ | $\begin{gathered} \text { Evap. } \\ h_{\mathrm{fg}} \end{gathered}$ | $\begin{gathered} \text { Sat. } \\ \text { Vapour } \\ h_{\mathrm{g}} \end{gathered}$ | Sat. Liquid sf | $\begin{gathered} \text { Evap. } \\ s_{f_{\mathrm{g}}} \end{gathered}$ | $\begin{gathered} \text { Sat. } \\ \text { Vapour } \\ s_{\mathrm{g}} . \end{gathered}$ |
| 105 | 0.12082 | 0.001047 | 1.4194 | $440.0{ }^{\circ}$ | 2072.3 | 2512.3 | 440.13 | 2243.7 | 2683.8 | 1.3629 | 5.9328 | 7.2958 |
| 110 | 0.14328 | 0.001052 | 1.2102 | 461.12 | 2057.0 | 2518.1 | 461.27 | 2230.2 | 2691.5 | 1.4184 | 5.8202 | 7.2386 |
| 115 | 0.16906 | 0.001056 | 1.0366 | 482.28 | 2041.4 | 2523.7 | 482.46 | 2216.5 | 2699.0 | 1.4733 | 5.7100 | 7.1832 |
| 120 | 0.19853 | 0.001060 | 0.8919 | 503.48 | 2025:8 | 2529.2 | 503.69 | 2202.6 | 2706.3 | 1.5275 | 5.6020 | 7.1295 |
| 125 | 0.2321 | 0.001065 | 0.77059 | 524.72 | 2009:9 | 2534.6 | 524.96 | 2188.5 | 2713.5 | 1.5812 | 5.4962 | 7.0774 |
| 130 | 0.2701 | 0.001070 | 0.66850 | 546.00 | 1993.9 | 2539.9 | 546.29 | 2174.2 | 2720.5 | 1.6343 | 5.3925 | 7.0269 |
| 135 | 0.3130 | 0.001075 | 0.58217 | 567.34 | 1977.7 | 2545.0 | 567.67 | 2159.6 | 2727.3 | 1.6869 | 5.2907 | 6.9777 |
| 140 | 0.3613 | 0.001080 | 0.50885 | 588.72 | 1961.3 | 2550.0 | 589.11 | 2144.8 | 2733.9 | 1.7390 | 5.1908 | 6.9298 |
| 145 | 0.4154 | 0,001085 | 0.44632 | 610.16 | 1944.7 | 2554.9 | 610.61 | 2129.6 | 2740.3 | 1.7906 | 5.0926 | 6.8832 |
| 150 | 0.4759 | 0.001090 | 0.39278 | 631.66 | 1927.9 | 2559.5 | 632.18 | 2114.3 | 2746.4 | 1.8417 | 4.9960 | 6.8378 |
| 155 | 0.5431 | 0.001096 | 0.34676 | 653.23 | 1910.8 | 2564.0 | 653.82 | 2098.6 | 2752.4 | 1.8924 | 4.9010 | 6.7934 |
| 160 | 0.6178 | 0.001102 | 0.30706 | 674.85 | 1893.5 | 2568.4 | 675.53 | 2082.6 | 2758.1 | 1.9426 | 4.8075 | 6.7501 |
| 165 | 0.7005 | 0.001108 | 0.27269 | 696.55 | 1876.0 | 2572.5 | 697:32 | 2066.2 | 2763.5 | 1.9924 | 4.7153 | 6.7078 |
| 170 | 0.7917 | 0.001114 | 0.24283 | 718.31 | 1858.1 | 2576.5 | 719.20 | 2049.5 | 2768.7 | 2.0418 | 4.6244 | 6.6663 |
| 175 | 0.8920 | 0.001121 | 0.21680 | 740.16 | 1840.0 | 2580.2 | 741.16 | 2032.4 | 2773.6 | 2.0909 | 4.5347 | 6.6256 |
| 180 | 1.0022 | 0.001127 | 0.19405 | 762.08 | 1821.6 | 2583.7 | 763.21 | 2015.0 | 2778.2 | 2.1395 | 4.4461 | 6.5857 |
| 185 | 1.1227 | 0.001134 | 0.17409 | 784.08 | 1802.9 | 2587.0 | 785.36 | 1997.1 | 2782.4 | 2.1878 | 4.3586 | 6.5464 |
| 190 | 1.2544 | 0.001141 | 0.15654 | 806.17 | 1783.8 | 2590.0 | 807.61 | 1978.8 | 2786.4 | 2.2358 | 4.2720 | 6.5078 |
| 195 | 1.3978 | 0.001149 | 0.14105 | 828.36 | 1764.4 | 2592.8 | 829.96 | 1960.0 | 2790.0 | 2.2835 | 4.1863 | 6.4697 |
| 200 | 1.5538 | 0.001156 | 0.12736 | 850.64 | 1744.7 | 2595.3 | 852.43 | 1940.7 | 2793.2 | 2.3308 2.3779 | 4.1014 | 6.4322 6.3951 |
| 205 | 1.7230 | 0.001164 | 0.11521 | 873.02 | 1724.5 | 2597.5 | 875.03 897.75 | 1921.0 | 2796.0 2798.5 | 2.3779 2.4247 | 4.0172 3.9337 | 6.3951 6.3584 |
| 210 | 1.9063 | 0.001173 | 0.10441 | 895.51 |  |  | 920.61 | 1879.9 | 2800.5 | 2.4713 | 3.8507 | 6.3221 |
| 215 | 2.1042 | 0.001181 | 0.09479 | 918.12 940.85 | 1682.9 1661.5 | $\begin{aligned} & 2601.1 \\ & 2602.3 \end{aligned}$ | 943.61 | 1858.5 | 2802.1 | 2.5177 | 3.7683 | 6.2860 |
| 220 | 2.3178 | 0.001190 | 0.08619 | 940.85 | 1661.5 | 2602.3 |  |  |  |  |  |  |

Steam Table

| $\begin{gathered} \text { Pressure } \\ \underset{P}{\text { MPa }} \end{gathered}$ | $\begin{aligned} & \text { Temp. } \\ & { }^{\text {ecp }} \end{aligned}$ | Specific Volume, $\mathrm{m}^{3} / \mathrm{kg}$ |  | Internal Energy, k/kg |  |  | Enthat |  |  | Entropy, k/kg-K |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $\begin{gathered} \text { Sat } \\ \substack{\text { Liquid } \\ v_{f}} \end{gathered}$ | $\begin{gathered} \text { Sat. } \\ \begin{array}{c} \text { Sapour } \\ v_{g} \end{array} \end{gathered}$ | $\begin{gathered} \text { Sat } \\ \substack{\text { Siquid } \\ u_{f}} \end{gathered}$ | $\begin{gathered} \text { Evap. } \\ u_{f g} \end{gathered}$ | $\begin{gathered} \text { Sat } \\ \text { Vapour }^{\prime} \\ u_{8} \end{gathered}$ | $\begin{gathered} \text { Sat } \\ \text { Liquid } \\ h_{f} \end{gathered}$ | $\begin{gathered} \text { Evap, } \\ h_{f(g} \end{gathered}$ | $\begin{gathered} \text { Sat. } \\ \left.\begin{array}{c} \text { Vapour } \\ h_{g} \end{array}\right) \end{gathered}$ | $\begin{gathered} \text { Sat } \\ \substack{\text { Liquid } \\ s_{\mathrm{f}}} \end{gathered}$ | $\underset{s_{\mathrm{fg}}}{\mathrm{Evap}^{2}}$ | $\begin{gathered} \text { Sat. } \\ \substack{\text { Vapour } \\ s_{g}} \end{gathered}$ |
| 0.275 | 130.60 | 0.001070 | 0.6573 | 548.57 | 1992.0 | 2540.5 | 548.87 | 2172.4 | 2721.3 | 1.6407 | 5.3801 | 7.0208 |
| 0.300 | 133:55 | 0.001073 | 0.6058 | 56173 | 1982.4 | 2543.6 | 561.45 | 2163.9 | 2725.3 | 1.6717 | 5.3201 | 6.9918 |
| 0.325 | 13630 | 0.001076 | 0.5620 | 572.88 | 1973.5 | 2546.3 | 573:23 | 2155.8 | 2729.0 | 1.7005 | 5.2646 | 6.9651 |
| 0.350 | 138.88 | 0.001079 | $0: 5243$ | 583.93 | 1965.0 | 2548.9 | 584.31 | 2148.1 | 2732.4 | 1.7274 | 5.2130 | 6.940 |
| 0.375 | 141:32 | 0.001081 | 0.4914 | 594.38 | 1956.9 | 2255.13 | 594.79 | 2140.8 | 2735.6 | 1.7527 | 5.1647 | 6.9174 |
| 0.40 | 143.63 | 0.001084 | 0.4625 | 604.29 | 1949.3 | 2553.6 | 604/73 | 2133.8 | 2738.5 | 1.776 | 5.1193 | 6.8958 |
| 0.45 | 147.93 | 0.001088 | 0.4140 | 622.75 | 1934.9 | 2557.6 | 623.24 | 2120.7 | 2743.9 | 1.8206 | 5.0359 | 6.8565 |
| 0.50 | 151.86 | 0.001093 | $0: 3749$ | 639.66 | 1921.6 | 2561.2 | 640.21 | 2108.5 | 2748.7 | 1.8606 | 4.9606 | 6.8212 |
| 0.55 | 155:48 | 0.001097 | 0.3427 | 655.30 | 1909.2 | 2564.3 | 655.91 | 2097.0 | 2752.9 | 1:8972 | 4.8920 | 6.789 |
| 0.60 | 158.85 | 0.001101 | $0: 3157$ | 669.88 | 1897.5 | 2567.4 | 670:54 | 2086.3 | 2756.8 | 1.9311 | 4.8289 | 6.760 |
| 0.65 | 162:01 | $0: 001104$ | 0.2927 | 683:55 | 1886.5 | 2570.3 | 684.26 | 2076.0 | 2760.3 | 1.9627 | 4.7704 | 6.7330 |
| 0:70 | 164.97 | 0.001108 | $0: 2729$ | 696.43 | 1876.1 | 2572.5 | 697:20 | 2066.3 | 2763.5 | 1.9922 | 4.7158 | 6.7080 |
| 0.75 | 167.97 | 0.001111 | 0.2556 | 708:62 | 1866.1 | 2574.7 | 709.45 | 2057.0 | 2766.4 | 2.0199 | 4.6647 | 6.6846 |
| 0.80 | 170.43 | 0,001115 | 0.2404 | 720.20 | 1856.6 | 2576.8 | 72110 | 2048.0 | 2769.1 | 2.0461 | 4.6166 | 6.6627 |
| 0.85 | 172.96 | 0.001118 | 0.2270 | 731:25 | 1847.4 | 2578.7 | 732:20 | 2039.4 | 2771.6 | 2.0709 | 4.5711 | 6.6421 |
| 0.90 | 175.38 | 0,001121 | 0.2150 | 741.81 | 1838.7 | 2580.5 | 742:82 | 2031.1 | 2773.9 | 2.0946 | 4.5280 | 6.6225 |
| 0.95 | 177.69 | 0.001124 | 0.2042 | 7519.94 | 1830.2 | 2582.1 | 753.00 | 2023.1 | 2776.1 | 2.1171 | 4.4869 | 6.6040 |
| 1.00 | 179.91 | 0.00\%127 | 0.19444 | 761.67 | 1822.0 | 2583.6 | 762.79 | 2015.3 | 2778.1 | 2.1386 | 4.4478 | 6.5864 |
| 1.10 | 184:09 | 0.001133 | 0.17753 | 780.08 | 1806.3 | 2686.4 | 78132 | 2000.4 | 2781.7 | 2.1791 | 4.3744 | 6.5535 |
| 1.20 | 187.99 | 0.001139 | 0.16333 | 797.27 | 1791.6 | 2588.8 | 798.64 | 1986.2 | 2784.8 | 2.2165 | 4.3067 | 6.5233 |
| 1.30 | 191.64 | 0,001144 | 0.15125 | 813.42 | 1777.5 | 2590.9 | 814.91 | 1972.7 | 2787.6 | 2.2514 | 4.2438 | 6.4953 |
| 1.40 | 195,07 | 0.001149 | 0.14084 | 828.68 | 1764.1 | 2592.8 | 830.29 | 1959.7 | 2790.0 | 2.2842 | 4.1850 | 6.4692 |
| 1.50 | 198.32 | 0.0001154 | 0.13177 | 843.14 | 1751.3 | 2594.5 | 844.87 | 19473 | 2792.1 | 2.3150 | 4.2198 | 6.4448 |
| 1.75 | 205.76 | 0.001166 | 0.11349 | 876.44 | 1721.4 | 2597.8 | 878.48 | 1918.0 | 2796.4 | 2.3851 | 4.0044 | 6.3895 |
| 2.00 | 212.42 | 0.001177 | 0.09963 | 906.42 | 1693.8 | 2600.3 | 908.77 | 1890.7 | 2799.5 | 2.4473 | 3.8935 | 6.3408 |
| 2.25 | 218.45 | 0.001187 | 0.08875 | 933.81 | 1668.2 | 2602.0 | 936.48 | 1865.2 | 2801.7 | 2.5034 | 3.7938 | 6.29 |

4. (a) A truncated cone has top and bottom diameters of 10 cm and 20 cm respectively, and a height of 10 cm . Calculate the shape factor between the top surface and the side, and also the shape factor between the side and itself. Use the figure showing the radiation shape factor for radiation between two parallel coaxial disks:

(b) A Francis turbine supplied through an 8.0 m diameter penstock has the following particulars:

Output power $=65000 \mathrm{~kW}$
Speed $=150$ r.p.m .
Hydraulic efficiency $=90 \%$
Flow rate $=120 \mathrm{~m}^{3} / \mathrm{s}$
Mean diameter of turbine at entry $=5 \mathrm{~m}$
Mean blade height at entry $=1.5 \mathrm{~m}$
Entry diameter of draft tube $=4.5 \mathrm{~m}$
Velocity in tailrace $=2.5 \mathrm{~m} / \mathrm{s}$
The static pressure head in the penstock measured just before entry to the runner is 60 m . The point of measurement is 3.2 m above the level of the tailrace. The loss in the draft tube is equivalent to $30 \%$ of the velocity head at entry to it. The exit plane of the runner is 2 m above the tailrace and the flow leaves the runner without swirl. Calculate:
(i) The overall efficiency
(ii) The direction of flow relative to the runner at inlet
(iii) The pressure head at entry to draft tube
(c) Two containers are connected with a pipe having a closed valve. One container contains a 5 kg mixture of $62.5 \% \mathrm{CO}_{2}$ and $37.5 \% \mathrm{O}_{2}$ on a mole basis at $30^{\circ} \mathrm{Cand} 125 \mathrm{kPa}$. The second container contains 10 kg of $\mathrm{N}_{2}$ at $15^{\circ} \mathrm{C}$ and 200 kPa .

The valve in the pipe is opened and gases are allowed to mix. During themixing process, 100 kJ of heat energy is supplied to the combined tank. Determine the volume of the mixture and write an energy balance equation.
[Required property tables are attached]
[20 Marks]

Molar mass, gas constant, and critical-point properties

| Substance | Formula | Molar mass, $M \mathrm{~kg} / \mathrm{kmol}$ | Gas constant, $R \mathrm{~kJ} / \mathrm{kg} \cdot \mathrm{K}^{*}$ | Critical-point properties |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | Temperature, K | Pressure, MPa | Volume, $\mathrm{m}^{3} / \mathrm{kmol}$ |
| Air | - | 28.97 | 0.2870 | 132.5 | 3.77 |  |
| Ammonia | $\mathrm{NH}_{3}$ | 17.03 | 0.4882 | 405.5 | 11.28 | 0.0883 |
| Argon | Ar | 39.948 | 0.2081 | 151 | 11.28 4.86 | 0.0749 |
| Berzene | $\mathrm{C}_{6} \mathrm{H}_{6}$ | 78.115 | 0.1064 | 562 | 4.92 | 0.2603 |
| Bromine | $\mathrm{Br}_{2}$ | 159.808 | 0.0520 | 584 | 10.34 | 0.1355 |
| $n$-Butane | $\mathrm{C}_{4} \mathrm{H}_{10}$ | 58.124 | 0.1430 | 425.2 | 3.80 | 0.2547 |
| Carbon dioxide | $\mathrm{CO}_{2}$ | 44.01 | 0.1889 | 304.2 | 7.39 | 0.0943 |
| Carbon monoxide | CO | 28.011 | 0.2968 | 133 | 3.50 | 0.0930 |
| Carbon tetrachloride | $\mathrm{CCl}_{4}$ | 153.82 | 0.05405 | 556.4 | 4.56 | 0.2759 |
| Chiorine | $\mathrm{CH}_{2}$ | 70.906 119.38 | 0.1173 | 417. | 7.71 | 0.1242 |
| Dichlorodifluoromethane (R-12) | $\mathrm{CHCl}_{3}$ | 119.38 120.91 | 0.06964 | 536.6 | 5.47 | 0.2403 |
| Dichlorofluoromethane (R-21) | $\mathrm{CCl}_{2} \mathrm{~F}_{2}$ | 120.91 102.92 | 0.06876 | 384.7 | 4.01 | 0.2179 |
| Ethane | $\mathrm{C}_{2} \mathrm{H}$ | 30.070 | 0.276 | 451.7 305.5 | 5.17 | 0.1973 |
| Ethyl alcohol | $\mathrm{C}_{2} \mathrm{H}_{3} \mathrm{OH}$ | 46.07 | 0.1805 | 05.5 | 6.38 | 0.1480 |
| Ethylene | $\mathrm{C}_{2} \mathrm{H}_{4}$ | 28.054 | 0.2964 | 282.4 | 6.38 5.12 | 0.1673 0.1242 |
| Helium | He | 4.003 | 2.0769 | 5.3 | 0.23 | 0.0578 |
| n-Hexane | $\mathrm{C}_{6} \mathrm{H}_{34}$ | 86.179 | 0.09647 | 507.9 | 3.03 | 0.3677 |
| Hydrogen (normal) | $\mathrm{H}_{2}$ | 2.016 | 4.1240 | 33.3 | 1.30 | 0.0649 |
| Krypton <br> Methane | $\mathrm{Kr}^{\mathrm{CH}}$ | 83.80 | 0.09921 | 209.4 | 5.50 | 0.0924 |
| Methane Methyl alcohol | $\mathrm{CH}_{4}$ | 16.043 | 0.5182 | 191.1 | 4.64 | 0.0993 |
| Methyl chloride | $\mathrm{CH}_{3} \mathrm{OH}$ $\mathrm{CH}_{3} \mathrm{Cl}$ | 32.042 50.488 | 0.2595 | 513.2 | 7.95 | 0.1180 |
| Neon | Ne | 50.488 20.183 | 0.1647 0.4119 | 416.3 44.5 | 6.68 | 0.1430 |
| Nitrogen | $\mathrm{N}_{2}$ | 28.013 | 0.4119 0.2968 | 126 | 2.73 | 0.0417 |
| Nitrous oxide | $\mathrm{N}_{2} \mathrm{O}$ | 44.013 | 0.1889 | 126.2 309.7 | 3.39 7.27 | 0.0899 0.0961 |
| Oxygen | $\mathrm{O}_{2}$ | 31.999 | 0.2598 | 154.8 | 5.08 | 0.0780 |
| Propane | $\mathrm{C}_{3} \mathrm{H}_{3}$ | 44.097 | 0.1885 | 370 | 4.26 | 0.1998 |
| Propylene | $\mathrm{C}_{3} \mathrm{H}_{6}$ | 42.081 | 0.1976 | 365 | 4.62 | 0.1810 |
| Sulfur dioxide | $\mathrm{SO}_{2}$ | 64.063 | 0.1298 | 430.7 | 7.88 | 0.1217 |
| Tetrafluoroethane (R-134a) | $\mathrm{CF}_{3} \mathrm{CH}_{2} \mathrm{~F}$ | 102.03 137.37 | 0.08149 | 374.2 | 4.059 | 0.1993 |
| Trichlorofluoromethane (R-11) Water | $\mathrm{CCl}_{3} \mathrm{~F}$ $\mathrm{H}_{2} \mathrm{O}$ | 137.37 | 0.06052 | 471.2 | 4.38 | 0.2478 |
| Water Xenon | $\mathrm{H}_{2} \mathrm{O}$ | 18.015 | 0.4615 | 647.1 | 22.06 | 0.0560 |
| Xenon | Xe | 131.30 | 0.06332 | 289.8 | 5.88 | 0.1186 |

 and $M$ is the molar mase.
loeal-gas specific heats of various common gases

| Gas | Formula | Gas constant, $R$ $\mathrm{kl} / \mathrm{kg} \cdot \mathrm{K}$ | C, $\mathrm{kJ} / \mathrm{kg} \cdot \mathrm{K}$ | $c$ $\mathrm{k} / \mathrm{kg} \cdot \mathrm{K}$ | $k$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Air | - | 0.2870 | 1.005 | 0.718 | 1.400 |
| Argon | Ar | 0.2081 | 0.5203 | 0.3122 | 1.667 |
| Butane | $\mathrm{C}_{4} \mathrm{H}_{10}$ | 0.1433 | 1.7164 | 1.5734 | 1.091 |
| Carbon dioxide | $\mathrm{CO}_{2}$ | 0.1889 | 0.846 | 0.657 | 1.289 |
| Carbon monoxide | CO | 0.2968 | 1.040 | 0.744 | 1.400 |
| Ethane | $\mathrm{C}_{2} \mathrm{H}_{6}$ | 0.2765 | 1.7662 | 1.4897 | 1.186 |
| Ethylene | $\mathrm{C}_{2} \mathrm{H}_{4}$ | 0.2964 | 1.5482 | 1.2518 | 1.237 |
| Helium | He | 2.0769 | 5.1926 | 3.1156 | 1.667 |
| Hydrogen | $\mathrm{H}_{2}$ | 4.1240 | 14.307 | 10.183 | 1.405 |
| Methane | $\mathrm{CH}_{4}$ | 0.5182 | 2.2537 | 1.7354 | 1.299 |
| Neon | Ne | 0.4119 | 1.0299 | 0.6179 | 1.667 |
| Nitrogen | $\mathrm{N}_{2}$ | 0.2968 | 1.039 | 0.743 | 1.400 |
| Octane | $\mathrm{C}_{5} \mathrm{H}_{28}$ | 0.0729 | 1.7113 | 1.6385 | 1.044 |
| Oxygen | $\mathrm{O}_{2}$ | 0.2598 | 0.918 | 0.658 | 1.395 |
| Propane | $\mathrm{C}_{3} \mathrm{H}_{3}$ | 0.1885 | 1.6794 | 1.4909 | 1.126 |
| Steam | $\mathrm{H}_{2} \mathrm{O}$ | 0.4615 | 1.8723 | 1.4108 | 1.327 |

Note : The wnit $\mathbf{k} J / / \mathbf{k} \mathbf{K}$ is equivalent to $\mathbf{k J} / \mathbf{k g}{ }^{*} \mathrm{C}$.

## SECTION-B

5. (a) A six-cylinder SI engine operates on a four-stroke cycle. The bore of each cylinder is 75 mm and the stroke is 100 mm . The clearance volume percylinder is 60 cc . At a speed of $4000 \mathrm{r} . \mathrm{p} . \mathrm{m}$., the fuel consumption is $18 \mathrm{~kg} / \mathrm{h}$ and the torque developed is $140 \mathrm{~N}-\mathrm{m}$. Calculate the
(i) brake thermal efficiency;
(ii) relative efficiency on the basis of brake power.

The calorific value of the fuel can be taken as $45000 \mathrm{~kJ} / \mathrm{kg}$.
(b) Draw the T-s and h-s diagrams for steam jet refrigeration system and write the expressions for the following
(i) Nozzle efficiency
(ii) Entrainment efficiency
(iii) Compression efficiency
(c) Briefly describe a natural draught cooling tower. Explain why it is hyperbolic inshape.
(d) Distinguish among the following
(i) Renewable energy
(ii) Green energy
(iii) Clean energy

Also, mention the relative environmental effects of the above.
(e) Describe the emission norms for Indian vehicles if they have to comply withBharat Stage (BS) Emission Standards-VI. Mention the devices and technologyintroduced to meet the BS-VI norms.
[12 Marks]
6. (a) A gasoline engine has a stroke volume of $0.002 \mathrm{~m}^{3}$ and a compression ratio of 6 . At the end of the compression stroke, the pressure is 10 bar and the temperature is $400^{\circ} \mathrm{C}$. Ignition is set so that the pressure rises along a straight line during combustion and attains its highest value of 30 bar after the piston has travelled $(1 / 40)$ of the stroke. The charge consists of a gasoline-air mixture in proportion of 1: 18 by mass. Calculate the heat lost per kg of charge during combustion. Take $\mathrm{R}=287 \mathrm{~J} / \mathrm{kg}-\mathrm{K}$, calorific value of the fuel $=45 \mathrm{MJ} / \mathrm{kg}, \mathrm{C}_{\mathrm{p}}=1 \mathrm{~kJ} / \mathrm{kg}$.
(b) A room is designed for air conditioning as per the following data

Room sensible heat gain $=30 \mathrm{~kW}$
Room latent heat gain $=10 \mathrm{~kW}$
Inside design conditions are : $25^{\circ} \mathrm{C}$ DBT and $50 \% \mathrm{RH}$
Outside conditions are : $40^{\circ} \mathrm{C}$ DBT and $27^{\circ} \mathrm{C}$ WBT
Bypass factor of the cooling coil $=0.10$
The return air from the space is mixed with the outside air before entering the cooling coil in the ratio of 4: 1 by weight. Determine the following:
(i) Apparatus dew point
(ii) Condition of air leaving the cooling coil
(iii) Quantity of dehumidified air
(iv) Mass of ventilation air
(v) Volume flow rate of fresh air
(vi) Total refrigeration load
[Psychrometric chart is attached]


Ref. Point for SHF is $25^{\circ} \mathrm{C}, 50 \% \mathrm{RH}$
(c) The angles at inlet and discharge of the blading of a $50 \%$ reaction turbine are $35^{\circ}$ and $20^{\circ}$ respectively. The speed of rotation is 1500 r.p.m. and at a particular stage, the mean ring diameter is 0.67 m and the steam condition is at 1.5 bar, 0.96 dry. Determine-
(i) the required height of blading to pass $3.6 \mathrm{~kg} / \mathrm{s}$ of steam;
(ii) the power developed by the ring.
[Saturated steam table is attached at the end]
7. (a) The following data refer to a boiler unit consisting of an economizer, a boiler and a superheater:

Mass of water evaporated per hour $=5940 \mathrm{~kg}$
Mass of coal burnt per hour $=675 \mathrm{~kg}$
Lower calorific value of coal $=31600 \mathrm{~kJ} / \mathrm{kg}$
Pressure of steam at boiler stop valve $=14$ bar
Temperature of feedwater entering economizer $=32^{\circ} \mathrm{C}$
Temperature of feedwater leaving economizer $=115^{\circ} \mathrm{C}$
Dryness fraction of steam leaving boiler and entering superheater $=0.96$
Temperature of steam leaving superheater $=260^{\circ} \mathrm{C}$
Specific heat of superheater steam $=2.3 \mathrm{~kJ} / \mathrm{kgK}$
Determine the following:
(i) Percentage of heat in coal utilized in economizer, boiler and superheater
(ii) Overall efficiency of the boiler unit

Assume specific heat of water $=4.187 \mathrm{~kJ} / \mathrm{kg}-\mathrm{K}$
[Saturated steam table is attached at the end]
(b) (i) Explain the various factors affecting anaerobic digestion process. Why do anaerobic microbes normally grow at a much lower rate than aerobicbacteria?
(ii) A family biogas plant is required to be designed to utilize the cow dung of five cows. The hydraulic retention time is 30 days. The temperature of the digester is to be maintained at $30^{\circ} \mathrm{C}$. The dry matter consumption per day is 2 kg . The biogas yield is $0.25 \mathrm{~m}^{3} / \mathrm{kg}$. The efficiency of the burner is $60 \%$. The heat of combustion of methane is $26 \mathrm{MJ} / \mathrm{m}^{3}$. The methaneproportion is $70 \%$. The density of feedstock material may be taken as $50 \mathrm{~kg} / \mathrm{m}^{3}$. Find (1) the volume of biogas digester and (2) its thermal power.
[10 + 10 Marks]
(c) (i) A refrigeration system with R-22 as refrigerant operates with an evaporating temperature of $-10^{\circ} \mathrm{C}$ and a condensing temperature of $35^{\circ} \mathrm{C}$. If the vapour leaves the evaporator saturated and is compressed isentropically,what is the COP of the cycle--(1) if saturated liquid enters the expansion device and (2) if the refrigerant entering the expansion device is with $10 \%$ vapour?
[ R -22 refrigerant chart is attached]

(ii) What is a liquid-to-suction heat exchanger in refrigeration and air conditioning? Illustrate the benefits of liquid-to-suction heat exchanger.
[10 Marks]
8. (a) (i) Describe the working principle of hydrogen fuel cell. Also, comment on the reversible energy conversion efficiency of fuel cells.
(ii) A flat plate solar collector measuring $2 \mathrm{~m} \times 1.2 \mathrm{mhas}$ a loss resistance of $0.13 \mathrm{~m}^{2} \mathrm{~K} / \mathrm{W}$ and a plate transfer efficiency of 0.85 . The glass cover has transmittance of 0.9 and the absorptance of the plate is also 0.9 . Water enters at a temperature of $35^{\circ} \mathrm{C}$. The ambient temperature is $20^{\circ} \mathrm{C}$ and the irradiance in the plane of the collector is $750 \mathrm{~W} / \mathrm{m}^{2}$. Calculate the flow rate needed to produce a temperature rise of $10^{\circ} \mathrm{C}$. The density of water and its specific heat at mean film temperature may be taken as $1000 \mathrm{~kg} / \mathrm{m}^{3}$ and $4.2 \mathrm{~J} / \mathrm{g}-{ }^{\circ} \mathrm{C}$ respectively.
[10 Marks]
(b) A two-pass surface condenser is required to handle the exhaust from a turbine developing 15 MW with specific steam consumption of $5 \mathrm{~kg} / \mathrm{kWh}$. The condenser vacuum is 660 mm of mercury when the barometer reads 760 mm of mercury. The mean velocity of water is $3 \mathrm{~m} / \mathrm{s}$ and the water inlet temperature is $24^{\circ} \mathrm{C}$. The condensate is saturated water and the outlet temperature of cooling water is $4^{\circ} \mathrm{C}$ less than the condensate temperature. The quality of exhaust steam is 0.9 dry. The overall heat transfer coefficient based on outer area of tubes is $4000 \mathrm{~W} / \mathrm{m}^{2}-{ }^{\circ} \mathrm{C}$. The water tubes are 38.4 mm in outer diameter and 29.6 mm in inner diameter. Calculate the following:
(i) Mass of cooling water circulated in $\mathrm{kg} / \mathrm{min}$
(ii) Condenser surface area
(iii) Number of tubes required per pass
(iv) Tube length

Assume atmospheric pressure to be 760 mm of mercury or 1.01325 bar and specific heat of water $=4.187 \mathrm{~kJ} / \mathrm{kg}-\mathrm{K}$.
[Saturated steam table is attached at the end]
(c) The total pressure maintained in an Electrolux refrigerator is 15 bar. Thetemperature obtained in the evaporator is $-15^{\circ} \mathrm{C}$. The quantities of heatsupplied to the generator are
(i) 420 kJ to dissociate one kg of vapourand
(ii) $1460 \mathrm{~kJ} / \mathrm{kg}$ for increasing the total enthalpy of $\mathrm{NH}_{3}$. The enthalpy of $\mathrm{NH}_{3}$ entering the evaporator is $330 \mathrm{~kJ} / \mathrm{kg}$. Take the following properties of $\mathrm{NH}_{3}$ at $-15^{\circ} \mathrm{C}$ :

Pressure $=2.45$ bar
Enthalpy of vapour $=1666 \mathrm{~kJ} / \mathrm{kg}$
Specific volume $=0.5 \mathrm{~m}^{3} / \mathrm{kg}$
The hydrogen enters the evaporator at $25^{\circ} \mathrm{C}$
Gas constant for $\mathrm{H}_{2}=4.218 \mathrm{~kJ} / \mathrm{kg} .{ }^{\circ} \mathrm{C}$
$\mathrm{C}_{\mathrm{p}}\left(\right.$ for $\left.\mathrm{H}_{2}\right)=12.77 \mathrm{~kJ} / \mathrm{kg}^{\circ} \mathrm{C}$
Find the COP of the system assuming $\mathrm{NH}_{3}$ leaves the evaporator in saturated condition.
[20 Marks]

## Saturated Steam Pressure Table

|  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |

Saturated Steam Pressure Table

| $\begin{gathered} p \\ \text { bar } \end{gathered}$ | ${ }^{\circ} \mathrm{C}$ | $\begin{gathered} v_{\mathrm{f}} \\ \mathrm{~m}^{3} / \mathrm{kg} \end{gathered}$ | $\begin{gathered} v_{g} \\ \mathrm{~m}^{3} / \mathrm{kg} \end{gathered}$ | $h_{f}$ <br> $\mathrm{kJ} / \mathrm{kg}$ | $h_{\mathrm{g}}$ <br> $\mathrm{kJ} / \mathrm{kg}$ | $\begin{gathered} h_{f g} \\ \mathrm{~kJ} / \mathrm{kg} \end{gathered}$ | $\begin{gathered} s_{f} \\ \mathrm{~kJ} / \mathrm{kg}-\mathrm{K} \end{gathered}$ | $\begin{gathered} 5_{8}^{8} \\ \mathrm{~kJ} / \mathrm{kg}-\mathrm{K} \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 20 | 120.23 | . 0010008 | 88544 | 504.70 | 27043 | 2201.6 |  |  |
| 2.1 | 121.78 | . 0010623 | 84500 | 511.28 | 2708.5 | 2201.6 | 1.5301 | 7.1268 |
| 22 | 123.27 | . 0010615 | .80084 | 517.62 | 2710.6 | 2193.0 | 1.5627 | 7.1106 |
| 2.3 | 124.71 | .0010650 | .77681 | 523.73 | 2712.6 | 2188.0 | 1.6827 | 7.0049 7.0800 |
| 2.4 | 125.99 | .00100e3 | .74645 | 529.63 | 2714.5 | 2184 | 1.5929 | 7.08067 |
| 2.5 | 127.43 | .0010675 | 71844 | 535.34 | 27164 | 2181.0 | 1.6071 |  |
| 2.8 | 128.73 | .0010888 | . 00251 | 540.87 | 2718.2 | 2177.3 | 1.6209 | $\begin{aligned} & 7.0620 \\ & 7039 \end{aligned}$ |
| 2.7 | 129.98 | .0010700 | . 66844 | 546.24 | 2718.9 | 2173.6 | 1.6209 1.6342 | 7.0389 7.0288 |
| 2.8 | 131.20 | .0010712 | 54604 | 851.44 | 2721.5 | 2170.1 | 1.6471 | 7.0262 |
| 2.9 | 132.39 | . 0010724 | . 62513 | \$56.50 | 2723.1 | 2168.8 | 1.6471 | 7.0140 $70023$ |
| 3.0 | 133.54 | . 0010735 | . 60565 | 561.43 | 2724.7 |  |  |  |
| 3.1 | 134.68 | . 0010746 | . 5872 | 868.23 | 2726.1 | 2163.2 | 1.6716 1.684 | 6.9809 8.9709 |
| 3.2 | 135.75 | . 0010787 | . 56090 | 870.90 | 2727.6 | 2158.7 | 1.0048 | 6.979 |
| 3.3 | 13688 | . 0010768 | . 55376 | 575.46 | 27290 | 2153.5 | 1.7000 | 6.9589 |
| 3.4 | 137.86 | .0010779 | . 53846 | 579.92 | 2730.3 | 2150.4 | 1.7168 | 8.9499 |
| 3.5 | 138.87 | . 0010789 | . 52400 | 58.27 | 2731.6 | 2147.4 | 1.7273 |  |
| 3.6 | 139.80 | . 0010799 | . 51032 | 588.53 | 2732.9 | 2144.4 | 1.7376 | 6.9392 |
| 3.7 | 140.83 | . 0010809 | 49736 | 502.69 | 2734.1 | 2141.4 | 1.7476 | 6.9205 |
| 3.8 | 141.78 | . 0010619 | . 48505 | 506.76 | 2735.3 | 2138.6 | 1.7574 | 6.9116 |
| 3.9 | 142.71 | . 0010629 | 47336 | 600.76 | 2736.5 | 2135.7 | 1.7670 | 6.9028 |
| 4.0 | 143.62 | .0010839 | 40222 | 604.67 | 2737.6 | 2133.0 | 1.7764 |  |
| 4.1 | 144.52 | . 0010348 | 45162 | 608.51 | 2738.7 | 21302 | 1.7856 | 6.8960 |
| 4.2 | 145.39 | . 0010858 | . 415150 | 612.27 | 2739.8 | 2187.5 | 1.7945 | 6.8779 |
| 4.3 | 148.25 | . 0010867 | 13184 | 615.97 | 2740.9 | 2124.9 | 1.8033 | 8.8700 |
| 4.4 | 147.09 | . 0010876 | . 22260 | 618.60 | 2741.9 | 2122.3 | 1.8120 | 8.8623 |
| 4.5 | 147.92 | .0010685 | . 61375 | 623.16 | 2742.9 | 2119.7 |  |  |
| 4.6 | 148.73 | . 0010804 | 40528 | 826.67 | 2743.9 | 2117.2 | 1.8287 | 6.8847 |
| 4.7 | 149.53 | . 0010003 | . 39716 | 630.11 | 2744 | 2114.7 | 1.838 | 6.8401 |
| 4.8 | 150.31 | . 0010911 | 38936 | 633.50 | 2745.7 | 21122 | 1.8448 | 6.8330 |
| 4.9 | 151.08 | . 0010020 | .38188 | 636.83 | 2746.6 | 21098 | 1.5227 | $\begin{aligned} & 68330 \\ & 6.8200 \end{aligned}$ |
| 5.0 | 151.84 | . 0010928 | . 37468 | 640.12 |  |  | 1.8004 | 6.8192 |
| 5.2 | 153.33 | . 00109095 | . 36108 | 646.53 | 2749.3 | 2102.7 | 1.8754 | 6.8059 |
| 5.4 | 154.76 | . 0010961 | 34846 | 652.76 | 2750.9 | 2098.1 | 1.8800 | 6.7932 |
| 5.6 | 156.16 | . 0010977 | . 33671 | 658.81 | 2752.5 | 2003.7 | 1.9040 | 6.7809 |
| 5.8 | 157.52 | . 0010093 | . 32574 | 664.69 | 2754.0 | 2069.3 | 1.9176 | $\begin{aligned} & 6.7809 \\ & 6.7600 \end{aligned}$ |
| 6.0 | 158.84 | . 0011009 | 31847 | 670.42 | 2755.5 |  |  | 6.7575 |
| 62 | 160.12 | . 0011024 | 30885 | 676.01 | 2756.9 | 2080.9 | 1.9427 | 6.7464 |
| 64 | 161.38 | . 0011039 | 29681 | 631.46 | 2758.2 | 2076.8 | 1.9662 | 6.7357 |
| 6.6 | 162.60 | . 0011063 | 28830 | 688.78 | 2759.5 | 2072.7 | 1.9084 | 6.7357 6.722 |
| 68 | 163.79 | . 0011088 | 28027 | 601.98 | 2760.8 | 2068.8 | 1.1 .9802 | $\begin{aligned} & 6.7262 \\ & 6.7150 \end{aligned}$ |
| 7.0 | 164.96 | .0011082 | . 27268 |  |  |  | 1.9918 | 6.7062 |
| 7.2 | 166.10 | . 0011098 | . 28550 | 702.03 | $2763.2$ | 2061.1 | 2.0031 | 6.6966 |
| 74 | 167.21 | .0011110 | . 25870 | 708.90 | 2764 | 2037.4 | 2.1041 | 6.6862 |
| 7.6 | 168.30 | .0011123 | . 25224 | 711.67 | 2765.4 | 2063.7 | 2.0249 | 6.5771 |
| 7.8 | 160.37 | .0011137 | . 24610 | 716.35 | 2766.4 | 2050.1 | 2.0354 | 6.6883 |
| 80 | 170.61 | . 0011150 | 24026 | 720.94 | 2767.5 |  |  |  |
| 82 | 171.44 | . 0011163 | 23409 | 725.43 | 2768.5 | 2043.0 | 2.0467 2.0568 | 6.6506 6.6511 |
| 8.4 | 172.45 | . 0011176 | 22938 | 729.85 | 2769.4 | 2039.6 | 2.0657 | 6.6429 |
| 8.6 | 173.44 | .0011188 | . 22430 | 734.19 | 2770.4 | 2008.2 | 2.0657 | 6.6429 6.6348 |
| 8.8 | 174.41 | . 0011201 | . 21945 | 738.45 | 2771.3 | 2002.8 | 2.0763 2.0848 | 6.6348 <br> 6.6260 |
| 9.0 | 175.36 | .0011213 | . 21481 | 742.64 |  |  |  |  |
| 9.2 | 176.29 | .0011225 | . 21006 | 746.76 | 2772.1 | $\begin{aligned} & 2029.5 \\ & 2026.2 \end{aligned}$ | 2.0941 2.1093 | 6.6182 |
| 9.4 | 177.21 | .0011238 | .20610 | 750.82 | 2773.8 | 2023.0 | 2.1122 | 6.6116 6.6012 |
| 9.6 | 178.12 | . 0011250 | .20201 | 754.81 | 2774.6 | 2018.8 | 2.1210 | 6.6032 |
| 98 | 179.01 | .0011262 | . 19807 | 758.74 | 2775.4 | 2016.7 | 2.1297 | 6.5805 |
| 10.0 10.5 | 179.88 18208 | . 0011274 | . 19429 | 762.61 | 2776.2 | 2013.6 | 2.1382 | 6.5828 |
| 10.6 11.0 | 182.02 184.07 | . 0011308 | . 18545 | 772.03 | 2778.0 | 2005.9 | 2.1588 | 6.5659 |
| 11.5 | 18.07 | . 0011331 | .17738 | 781.12 | 2779.7 | 1908.5 | 2.1786 | 6.5497 |
| 12.0 | 187.96 | . 0011359 | . 16999 | 789.92 | 2781.3 | 19013 | 2.1977 | 6.5342 |
| 12.5 | 189.81 | . 00113812 | -16320 | 796.43 | 2782.7 | 1384.3 | 2.2161 | 6.5194 |
| 13.0 | 191.61 | . 0011433 | .15693 | 806.69 814.70 | 2784.1 | 1977.4 | 2.2338 | 6.5051 |
| 13.5 | 193.35 | . 0011464 | . 14574 | 814.70 822.49 | 27854 | 1970.7 | 2.2510 | 6.4913 |
| 14.0 | 195.04 | . 0011489 | . 14072 | 822.49 830.07 | 27866 27878 | 1964.2 | 2.2576 | 6.4760 |
| 14.5 | 196.69 | . 0011514 | . 13604 | 897.46 | 2787.8 2788.9 | 1957.7 1951.4 | 2.2837 2.2909 | 6.4651 6.4526 |

Saturated Steam Pressure Table

| $\begin{gathered} p \\ \text { bar } \end{gathered}$ | ${ }^{\circ} \mathrm{C}$ | $\begin{gathered} v_{f} \\ m^{3} / \mathrm{kg} \end{gathered}$ | $\begin{gathered} v_{g} \\ \mathrm{~m}^{3} / \mathrm{kg} \end{gathered}$ | $\begin{gathered} h_{f} \\ \mathrm{~kJ} / \mathrm{kg} \end{gathered}$ | $\begin{gathered} h_{\mathrm{g}} \\ \mathrm{~kJ} / \mathrm{kg} \end{gathered}$ | $\begin{gathered} h_{\mathrm{fg}} \\ \mathrm{~kJ} / \mathrm{kg} \end{gathered}$ | $\begin{gathered} s_{f} \\ \mathrm{~kJ} / \mathrm{kg}-\mathrm{K} \end{gathered}$ | $\stackrel{s_{\mathrm{g}}}{\mathrm{~kJ} / \mathrm{kg}-\mathrm{K}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 15.0 | 198.29 | . 0011539 | . 13166 | 844.65 | 2789.9 | 1945.2 | 23145 | 6.4406 |
| 15.5 | 199.85 | .0011563 | . 12785 | 851.60 | 2790.8 | 1939.2 | 23292 | 6.428 |
| 16.0 | 201.37 | .0011588 | . 12369 | 858.65 | 2791.7 | 19032 | 2.3436 | 6.4175 |
| 16.5 | 202.86 | . 0011610 | , 12005 | 865.28 | 2792.6 | 1927.3 | 2.3576 | 6.4065 |
| 17.0 | 204.31 | . 001163 | . 1162 | 871.84 | 2793.4 | 1921.5 | 2.3713 | 6.3957 |
| 17.5 | 206.72 | .0011656 | . 11338 | 878.27 | 27941 | 19159 | 2.3846 | 63853 |
| 18.0 | 207.11 | . 0011678 | . 11032 | 884.57 | 27948 | 19109 | 2.3976 | 6.3751 |
| 18.5 | 208.47 | . 0011701 | . 10741 | 800.75 | 2796.5 | 19047 | 2.4103 | 6.3651 |
| 19.0 | 200.80 | . 0011723 | . 10485 | 896.81 | 2796.1 | 18993 | 2.4228 | 6.3584 |
| 19.5 | 211.10 | . 0011744 | . 10203 | 902.75 | 2796.7 | 1883.9 | 2.4369 | 6.3459 |
| 20.0 | 212.37 | .0011766 | . 090953 | 908.59 | 2797.2 | 1888.6 | 2.4689 | 6.3367 |
| 20.5 | 213.63 | . 0011787 | . 097158 | 914.32 | 2797.7 | 18834 | 2.4585 | 6.3276 |
| 21.0 | 21485 | . 0011809 | . 094800 | 919.96 | 2798.2 | 1878.2 | 2.4700 | 6.3187 |
| 21.5 | 216.06 | .0011830 | .002723 | 925.50 | 2798.6 | 1873.1 | 2.4812 | 6.3100 |
| 22.0 | 21724 | . 0011850 | . 090052 | 930.95 | 2799.1 | 1868.1 | 2.4922 | 6.3015 |
| 22.5 | 218.41 | .0011871 | . 088660 | 936.32 | 2790.4 | 1863.1 | 2.5030 | 6.2931 |
| 23.0 | 219.55 | . 0011892 | . 086769 | 941.00 | 27998 | 1858.2 | 2.5136 | 62849 |
| 23.5 | 220.68 | . 0011912 | 064048 | 946.80 | 2800.1 | 1853.3 | 2.5241 | 62769 |
| 24.0 | 221.78 | .0011932 | . 083190 | 961.93 | 2800.4 | 1848.5 | 2.5343 | 63600 |
| 24.5 | 222.87 | .0011962 | .081520 | 956.98 | 2800.7 | 1843.7 | 2.514 | 6.2612 |
| 25.0 | 223.94 | .0011972 | . 079805 | 961.96 | 28009 | 1839.0 | 2.5543 | 6.2536 |
| 25.5 | 225.00 | . 0011901 | . 078352 | 906.87 | 2801.2 | 1834.3 | 2.5640 | 6.2461 |
| 26.0 | 226.04 | . 0012011 | .076856 | 971.72 | 2801.4 | 1829.6 | 2.5736 | 6.2387 |
| 26.5 | 227.06 | . 0012081 | .075615 | 976.50 | 2801.8 | 1825.1 | 2.5831 | 62315 |
| 27.0 | 228.07 | . 0012050 | . 074025 | 981.22 | 2801.7 | 1820.5 | 2.5924 | 6.2244 |
| 27.5 | 229.07 | .0012060 | . 072884 | 985.88 | 2801.9 | 1816.0 | 2.8016 | 6.2173 |
| 28.0 | 230.05 | . 0012088 | .071389 | 900.48 | 28020 | 1811.5 | 2.6108 | 6.2104 |
| 28.5 | 231.01 | . 0012107 | . 070138 | 995.03 | 2802.1 | 1807.1 | 2.6196 | 6.2036 6.1909 |
| 29.0 | 231.97 | . 0012126 | . 088928 | 909.52 | 2802.2 | 1802.6 | 2.6283 | 6.1969 6.1903 |
| 29.5 | 232.91 | . 0012145 | . 067758 | 1003.96 | 2802.2 | 1798.3 | 2.6370 | 6,1903 |
| 30.0 | 233.84 | .0012163 | .060626 | 1003.35 | 2802.3 | 1793.9 | 2.8455 | 6.1837 |
| 31.0 | 235.67 | . 0012200 | .064467 | 1016.99 | 2802.3 | 1785.4 | 2.6623 | 6.1709 |
| 32.0 | 237.45 | .0012237 | . 062439 | 1095.43 | 2802.3 | 17769 | 2.6786 | 6.1585 |
| 33.0 | 239.18 | . 0012274 | .060629 | 1093.70 | 2502.3 | 1768.6 | 2.8945 | 6.1463 |
| 34.0 | 240.88 | . 0012310 | . 068728 | 1041.81 | 2802.1 | 1760.3 | 2.7101 | 6.1344 |
| 35.0 | 242.54 | . 0012345 | . 057025 | 1049.76 | 2802.0 | 1752.2 | 2.7253 | 6.1228 |
| 36.0 | 24.16 | . 0012381 | . 065415 | 1067.56 | 2801.7 | 17442 | 2.7401 | 6.1115 |
| 37.0 | 245.75 | . 0012416 | . 063881 | 1065.21 | 28014 | 17362 | 2.7547 | 6.1004 |
| 38.0 | 247.31 | . 0012451 | . 062438 | 1072.74 | 2801.1 | 1728.4 | 2.7689 | 6.0696 |
| 39.0 | 24884 | .0012486 | . 051061 | 1080.13 | 2800.8 | 1720.6 | 27829 | 6.0769 |
| 40.0 | 250.33 | . 0012521 | . 049749 | 1087.40 | 2800.3 | 1712.9 | 2.7965 | 6.0685 |
| 40.1 | 251.80 | . 0012565 | . 048500 | 1004.56 | 2799.9 | 1705.3 | 2.8009 | 6.0583 |
| 42.0 | 253.24 | . 0012589 | . 047307 | 110160 | 2790.4 | 1697.8 | 2.8231 | 6.0482 6.0383 |
| 43.0 | 254.66 | . 0012623 | .046168 | 110854 | 2798.9 | 1690.3 | 2.8360 | 6.0383 6.0285 |
| 44.0 | 256.05 | . 0012657 | . 045079 | 1115.38 | 2798.3 | 1682.9 | 2.8487 | 6.0286 |
| 45.0 | 257.41 | . 0012691 | . 044007 | 1122.11 | 2797.7 | 1675.6 | 2.8612 | 6.0191 |
| 48.0 | 258.75 | . 0012725 | . 043038 | 1128.76 | 2797.0 | 16883 | 2.8735 | 6.0097 |
| 47.0 | 260.07 | . 0012758 | . 042081 | 1135.31 | 2796.4 | 1681.1 | 2.8855 | 8.0004 |
| 48.0 | 261.37 | . 0012792 | . 041161 | 1161.78 | 2795.7 | 1653.9 | 2.8974 | 5.9913 |
| 42.0 | 262.65 | . 0012825 | . 040278 | 1148.16 | 2794.9 | 16468 | 2.9091 | 5.9824 |
| 50.0 | 263.91 | .0012858 | . 039429 | 1154.47 | 2794.2 | 1639.7 | 2.9206 | 5.9735 |
| 51.0 | 205.15 | . 0012891 | . 038611 | 1160.69 | 2793.4 | 1632.7 | 2.9313 | 59648 |
| 52.0 | 206.37 | . 0012924 | . 037824 | 1166.85 | 2792.6 | 1625.7 16188 | 2.9431 | 5.9561 59476 |
| 53.0 | 257.58 | . 0012957 | . 037066 | 1172.93 | 2791.7 | 1618.8 | 2.9641 | 5.9476 59392 |
| 54.0 | 268.76 | . 0012990 | . 036334 | 1178.94 | 2790.8 | 1611.9 | 2.9650 | 5.9392 |
| 58.0 | 269.90 | . 0013023 | . 035628 | 118489 | 2780.9 | 1605.0 | 2.9757 | 5.9309 |
| 56.0 | 27.09 | . 0013066 | . 03496 | 1190.77 | 2789.0 | 1598.2 | 2.9863 | 5.9227 |
| 57.0 | 272.28 | . 0013069 | . 034288 | 1196.59 | 27880 | 1591.4 | 2.9968 | 5.9146 |
| 58.0 | 273.35 | . 0013121 | . 033651 | 1202.35 | 2787.0 | 1584.7 | 3.0071 | 5.9066 5.8998 |
| 50.0 | 274.46 | . 0013154 | . 033034 | 1208.05 | 2786.0 | 1578.0 | 3.0172 | 5.8986 |
| 60.0 | 275.55 | .0013187 | . 032438 | 1213.69 | 2785.0 | 1571.3 | 3.0273 | 5.8908 5.8830 |
| 61.0 | 276.63 | . 0013219 | . 031800 | 1219.28 | 2784.0 | 1564.7 | 3.0372 | 5.8830 |
| 62.0 | 277.70 | . 0013252 | . 031300 | 1224.82 | 2782.9 | 15588 | 3.0471 | 5.8763 |
| 63.0 | 278.75 | . 0013285 | . 030757 | 1230.31 | 2781.8 | 1561.5 | 3.0668 | 58677 58801 |
| 64.0 | 279.79 | . 0013317 | 030230 | 1235.75 | 2780.6 | 1544.9 | 3.0664 | 5.8601 |

